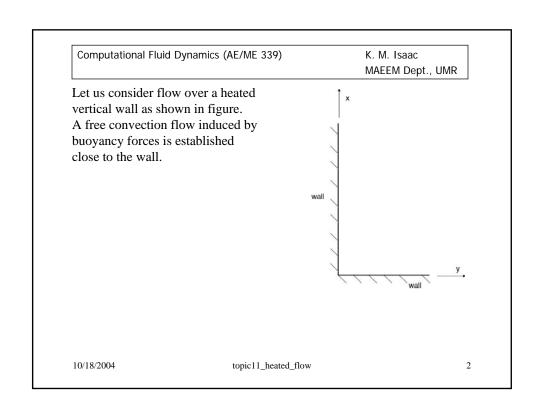
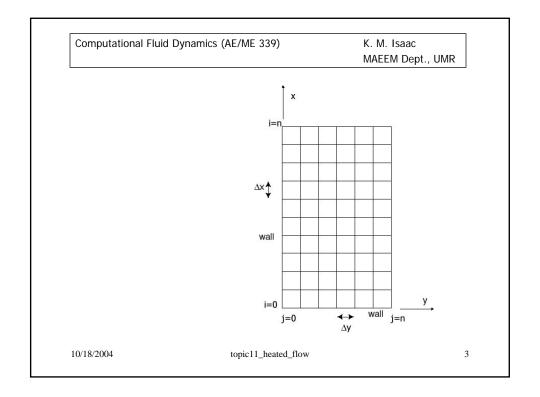


AE/ME 339 Computational Fluid Dynamics (CFD)

K. M. Isaac

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Boundary conditions:

At
$$t = 0$$
:
 $u = v = 0$, $T = T_{inf}$ (everywhere in the solution domain)

x i=n x i=n y i=n y i=n x i=n

At t > 0:

y=0: u=v=0, $T=T_w$ (heated wall condition y=inf: u=0, $T=T_{inf}$ (far field condition) x=0: u=v=0, $T=T_{inf}$ (bottom wall conditic

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Governing equations

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0....(1)$$

$$\frac{\partial u}{\partial t} + u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} = -\frac{1}{\rho} \frac{\partial p}{\partial x} + g_x \beta (T - T_{\infty}) + v \left(\frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} \right) \dots (2)$$

$$\frac{\partial v}{\partial t} + u \frac{\partial v}{\partial x} + v \frac{\partial v}{\partial y} = -\frac{1}{\rho} \frac{\partial p}{\partial y} + g_y \beta (T - T_{\infty}) + v \left(\frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial y^2} \right) \dots (3)$$

$$\frac{\partial T}{\partial t} + u \frac{\partial T}{\partial x} + v \frac{\partial T}{\partial y} = \frac{k}{\rho c_p} \left(\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} \right) + \frac{\mu}{\rho c_p} \left[2 \left(\frac{\partial u}{\partial x} \right)^2 + 2 \left(\frac{\partial v}{\partial y} \right)^2 + \left(\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} \right)^2 \right] \dots (4)$$

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The Navier-Stokes equations derived in Chapter 2 can be simplified using the boundary layer assumptions.

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0.....(1)$$

$$\frac{\partial u}{\partial t} + u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} = g_x \beta (T - T_{\infty}) + v \frac{\partial^2 u}{\partial y^2}.....(5)$$

$$\frac{\partial T}{\partial t} + u \frac{\partial T}{\partial x} + v \frac{\partial T}{\partial y} = \frac{k}{\rho c_p} \frac{\partial^2 T}{\partial y^2}$$

$$\frac{\partial T}{\partial t} + u \frac{\partial T}{\partial x} + v \frac{\partial T}{\partial y} = \alpha \frac{\partial^2 T}{\partial y^2}.....(6)$$

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Note the neglected terms in the above equations

- Eq. 2: 2nd derivative of u w.r.t x is neglected because it is small compared to the other term.
 Pressure gradient is assumed to be zero.
- Eq. 3: This equation is not solved in the present formulation All the terms on the RHS are assumed to be zero.

 Since p is assumed to be a constant, there is no need to use this equation.
- Eq. 4: 2nd derivative of temperature w.r.t x and the viscous dissipation terms are neglected.

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The simplified equations (continuity, x-momentum and energy) can be non-dimensionalized as follows:

$$\xi = x(g\beta\Delta T/v^2)^{\frac{1}{3}}$$

$$\eta = y(g\beta\Delta T/v^2)^{\frac{1}{3}}$$

$$\tau = t(g\beta\Delta T)^{\frac{2}{3}}/v^{\frac{1}{3}}$$

$$\overline{u} = u/(vg\beta\Delta T)^{\frac{1}{3}}$$

$$\overline{v} = v/(vg\beta\Delta T)^{\frac{1}{3}}$$

$$\theta = \frac{T - T_{\infty}}{T_1 - T_{\infty}} = \frac{T - T_{\infty}}{\Delta T}$$

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Note that gx is the magnitude of the gravity component in the x-direction. β is the volumetric coefficient of thermal expansion defined as

$$\beta = -\frac{1}{\rho} \frac{\partial \rho}{\partial T}$$

Further, we assume that the Prandtl number Pr (= μ cp/k= v/ α) = 1. The equations will now be obtained in a form that is easy to solve numerically.

Note β has units of (1/K)

If the assumption Pr=1 is not made, the RHS of the energy equation will have a Pr factor.

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The energy equation for non-unity Prandtl number can be differenced using the explicit formulation as follows

$$\frac{\theta_{i,j}^{n+1} - \theta_{i,j}^{n}}{\Delta \tau} + \overline{u}_{i,j}^{n} \frac{\theta_{i,j}^{n} - \theta_{i-1,j}^{n}}{\Delta \xi} + \overline{v}_{i,j}^{n} \frac{\theta_{i,j+1}^{n} - \theta_{i,j}^{n}}{\Delta \eta} = \frac{1}{\Pr} \left(\frac{\theta_{i,j+1}^{n} - 2\theta_{i,j}^{n} + \theta_{i,j-1}^{n}}{(\Delta \eta)^{2}} \right) \dots (7)$$

Note the use of BD for the second term on the LHS and FD for the third term and CD for the RHS term.

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The x-momentum equation can now be written in the finite difference form as

$$\frac{\overline{u}_{i,j}^{n+1} - \overline{u}_{i,j}^{n}}{\Delta \tau} + \overline{u}_{i,j}^{n} \frac{\overline{u}_{i,j}^{n} - \overline{u}_{i-1,j}^{n}}{\Delta \xi} + \overline{v}_{i,j}^{n} \frac{\overline{u}_{i,j+1}^{n} - \overline{u}_{i,j}^{n}}{\Delta \eta}$$

$$= \theta_{i,j}^{n+1} + \frac{\overline{u}_{i,j+1}^{n} - 2\overline{u}_{i,j}^{n} + \overline{u}_{i,j-1}^{n}}{\left(\Delta \eta\right)^{2}} \dots (8)$$

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The continuity equation becomes

$$\frac{\overline{u}_{i,j}^{n+1} - \overline{u}_{i-1,j}^{n+1}}{\Delta \xi} + \frac{\overline{v}_{i,j}^{n+1} - \overline{v}_{i,j-1}^{n+1}}{\Delta \eta} = 0....(9)$$

Equations 7, 8 and 9 are solved sequentially for each time step. Note that $\theta n+1$ is known once the energy equation is solved.

Therefore, in the momentum equation, it can be treated as a known quantity.

Since u is known from solving the x-momentum equation, vn+1 are the only unknowns in Eq. (9).

Thus these set of equations can be solved explicitly.

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