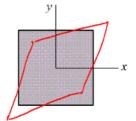
1. The strain components $\varepsilon_{x} = 520 \ \mu \varepsilon$, $\varepsilon_{y} = -650 \ \mu \varepsilon$, and $\gamma_{xy} = 750 \ \mu rad$ are given for a point in a body subjected to plane strain.

a. (4 points) Sketch the deformed shape on the element to the right.



b. (9 points) Determine the strain components ε_n , ε_n , and y_n at the point if the *nt*-axes are rotated with respect to the xy-axes by $\theta=35^\circ$, as illustrated in the figure. Do not sketch the deformed shape of the element.

$$\varepsilon_{n} = \frac{487.5}{2} \mu \varepsilon = \frac{520 - 650}{2} + \frac{520 + 650}{2} \cos 70 + \frac{950}{2} \sin 70$$

$$\varepsilon_{n} = \frac{-617.5}{2} \mu \varepsilon = \frac{11}{2} - \frac{11}{2} \cos 70 + \frac{750}{2} \cos 70$$

$$\varepsilon_{n} = \frac{-842.9}{2} \mu \cot = 2 \left[-\frac{520 + 650}{2} \sin 70 + \frac{750}{2} \cos 70 \right]$$

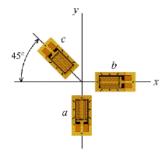
c. (14 points) Determine the angles θ_p and θ_s , principal strains, and maximum in-plane shear strain at the point. Do not sketch the deformed shapes of the element.

$$\theta_{p} = \frac{16.33}{\text{deg}} = \frac{1}{2} \tan^{-1} \left(\frac{750}{520 + 650} \right) \\
\theta_{s} = \frac{-28.67}{\text{deg}} = \frac{1}{2} \cot^{-1} \left(\frac{750}{520 + 650} \right) \\
\theta_{s} = \frac{629.9}{2} \mu \epsilon$$

$$\theta_{s} = \frac{629.9}{2} \mu \epsilon$$

$$\theta_$$

2. (10 points) The strain rosette shown in the figure was used to obtain normal strain data at a point on the free surface of a machine part. $\varepsilon_a = 550\mu$, $\varepsilon_b = -730\mu$, and $\varepsilon_c = -375\mu$. Determine the strain components ε_x , ε_y , and γ_w at the point.



$$\varepsilon_{x} = -730$$
 $\mu\epsilon$

$$y_{xy} = 570$$
 µrad

$$-375 = -\frac{730 + 550}{2} - \frac{730 - 550}{2} = \frac{0}{2} + \frac{1}{2} \times \frac{1}{2} = \frac{1}{2} \times \frac{1}{2} = \frac{1}{2} \times \frac{1}{2} = \frac{1}{2} \times \frac{1}{2} \times \frac{1}{2} = \frac{1}{2} \times \frac{1}{2} = \frac{1}{2} \times \frac{1}{2} = \frac{1}{2} \times \frac{1}{2} \times \frac{1}{2} \times \frac{1}{2} = \frac{1}{2} \times \frac{1}{2} \times \frac{1}{2} \times \frac{1}{2} \times \frac{1}{2} = \frac{1}{2} \times \frac{1}{2} \times$$

3. (8 points) The strain components ε_x = 390 μ, ε_y = 820 μ, and γ_{xy} = -560 μ are given for a point on the free surface of a machine component. The modulus of elasticity for the material is E = 73 GPa and the Poisson's ratio is v = 0.30. Determine the stresses σ_x and τ_{xy} at the point.

$$\sigma_{x} = \frac{51.02}{10.32} \text{ MPa} = \frac{73E9}{10.32} (390E-6+.3(820E-6))$$

$$\tau_{yy} = \frac{-15.72}{2(1+.3)} \text{ MPa} = \frac{73E9}{2(1+.3)} (-560E-6)$$

4. (5 points) A spherical gas-storage tank with an inside diameter of 12 m is being constructed to store gas under an internal pressure of 1.75 MPa. The tank will be constructed from structural steel that has a yield strength of 250 MPa. If a factor of safety of 3.0 with respect to the yield strength is required, determine the minimum wall thickness required for the spherical tank.

$$t_{min} = 63 \text{ mm}$$

$$C = \frac{1.75E6(6)}{2t} = \frac{250E6}{3}$$

5. (5 points) A cylindrical boiler with an outside diameter of 3.60 m and a wall thickness of 40 mm is made of a steel alloy that has a yield stress of 415 MPa. Determine the maximum normal stress produced by an internal pressure of 2 MPa.

$$\sigma_{max} = 88$$
 MPa = $\frac{2\bar{c}6(1.76)}{.04}$

6. A closed cylindrical vessel contains a fluid at a pressure of 720 psi. Assume $\sigma_{hoop} = 22.32$ ksi and $\sigma_{hool} = 11.16$ ksi. Determine:



 $a.\ (4\ points)$ the absolute maximum shear stress on the outer surface of the cylinder.

b. (4 points) the absolute maximum shear stress on the inner surface of the cylinder.

- 7. (9 points) Three loads are applied to the short rectangular post. The cross-sectional dimensions of the post are shown.
 - a. Using the positive sign convention shown, determine the following internal forces and moments acting on a x-z plane through points H and K.

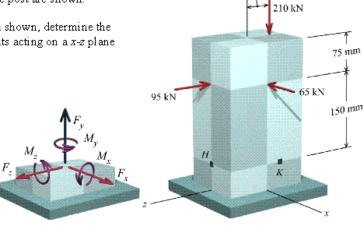
$$F_{x} = \frac{-65}{\text{kN}} \text{kN}$$

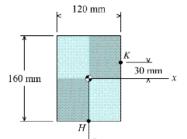
$$F_{y} = \frac{-210}{\text{kN}} \text{kN}$$

$$F_{z} = \frac{-95}{\text{kN}} \text{kN}$$

$$M_{x} = \frac{-24.75}{\text{kN-m}} \text{kN-m}$$

$$M_{y} = \frac{0}{\text{kN-m}} \text{kN-m}$$

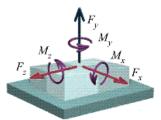




~ 50 mm

b. (14 points) For the following loadings, determine the normal and shear stresses at point ${\cal H}.$

$$\begin{split} F_x &= 48 \text{ kN} \\ F_y &= 0 \text{ kN} \\ F_z &= 73 \text{ kN} \\ M_x &= 3 \text{ kN-m} \\ M_y &= 0 \text{ kN-m} \\ M_z &= -2.5 \text{ kN-m} \end{split}$$



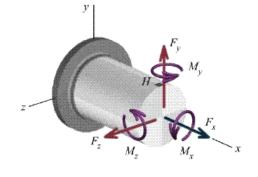
$$\sigma_{x} = \frac{D}{MPa}$$

$$\sigma_{y} = \frac{-6.859}{3.75} MPa = \frac{-3000 (.08)}{.(2(.16)^{3}/12)}$$

$$\tau_{xy} = \frac{3.75}{MPa} MPa = \frac{48,000 (.03)(.06)(.16)}{\frac{.(6(.12)^{3}}{12}(.16)}$$

 (14 points) A solid steel crank has an outside diameter of 30 mm. For the following loadings, determine the normal and shear stresses on the top surface of the crank at point H.

$$\begin{aligned} F_x &= 2350 \text{ N} \\ F_y &= -1275 \text{ N} \\ F_z &= 0 \text{ N} \\ M_x &= 204 \text{ Nm} \\ M_y &= 376 \text{ Nm} \\ M_z &= 0 \text{ Nm} \end{aligned}$$



$$\sigma_{x} = 3.325$$
 $MPa = \frac{2350}{47(.03)^{2}}$
 $\sigma_{z} = 0$
 MPa

$$\tau_{xz} = -38.48$$
 $MPa = \frac{-204(.015)}{32(.03)^{4}}$

